#### Feedback control

J.B. Burl. Linear Optimal Control:  $\mathcal{H}_2$  and  $\mathcal{H}_\infty$  methods. Addison-Wesley, 1998.

### Outline

- State-feedback
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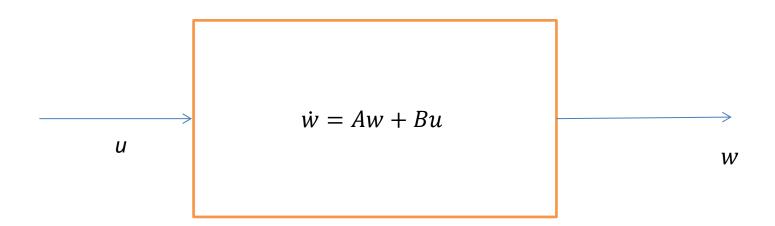
## State feedback Physical system

Reduced-order model of dynamics:

$$\dot{w} = Aw + Bu$$

Input: *u* 

Output: w



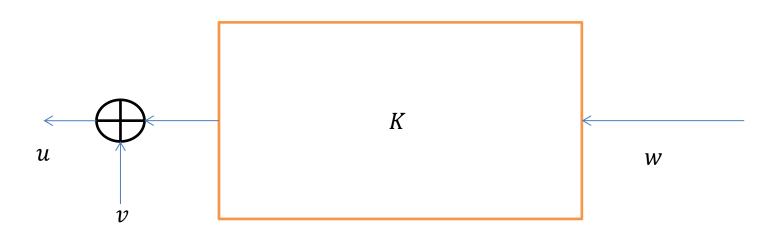
### State feedback Controller

The **controller** is a system that takes the state w and provides a control law u:

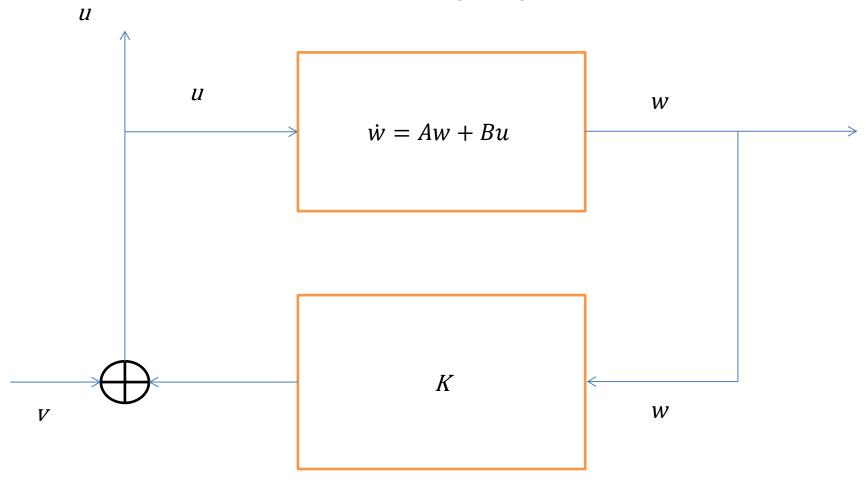
$$u = Kw + v$$

Inputs: the state w and the actuator noise v

Output: the control law u



## State feedback Closed-loop system



## State feedback Stability

Governing equation with **state feedback**:

$$\begin{cases} \dot{w} = Aw + Bu \\ u = Kw + v \text{ (Noisy actuator)} \end{cases}$$

Input: actuator noise v.

Hence:

$$\dot{w} = Aw + BKw + Bv = (A + BK)w + Bv$$

K is chosen so that A + BK is stable.

$$w(t) = \int_0^t e^{(A+BK)(t-\tau)} Bv(\tau) d\tau$$

### State feedback Performance

If performance is assessed by the measurement y = Cw, then:

$$y(t) = \int_0^t Ce^{(A+BK)(t-\tau)} Bv(\tau) d\tau.$$

In presence of white-noise v, the standard deviation of y,  $\sqrt{E(y^2)}$ , is proportional to the 2-norm of the closed-loop impulse function:  $\|\mathbf{Z}^{\text{cl}}(\mathbf{t})\|_2 = \sqrt{\int_0^\infty |Z^{cl}(t)|^2 dt}$ , where  $\mathbf{Z}^{\text{cl}}(\mathbf{t}) = Ce^{(A+BK)t}B$ .

Standard deviation of an output signal. Let us consider a stable system:

$$\dot{w} = A'w + B'v$$
$$y = C'w$$

If *v* is white-noise characterized by a PSD (Power Spectral Density) *S*, then the standard deviation of the output *y* is equal to:

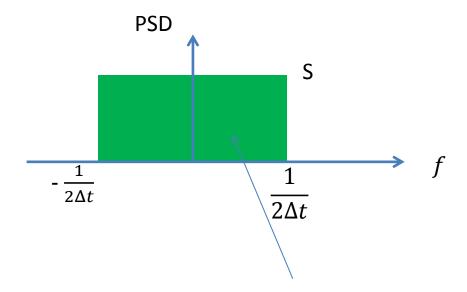
$$\sqrt{E(y^2)} = ||Z'(t)||_2 \sqrt{S}$$

where  $Z'(t) = C'e^{A't}B'$  is the impulse response of the system.

### State feedback Performance

Link between PSD, sampling time and variance of white noise: If  $\Delta t$  is the sampling time, then the variance of the white-noise is:

$$E(v^2) = \frac{S}{\Delta t}$$



Variance of signal is green area (Parseval)

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## Observer feedback Physical system

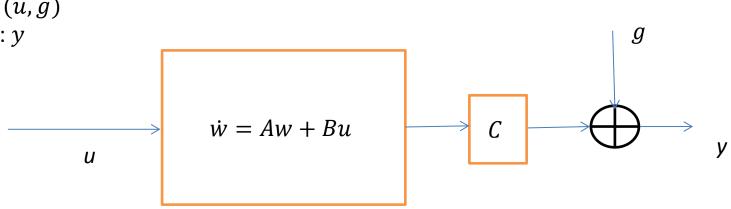
In fluid systems, state feedback is not realistic since w is unknown. Can we apply K on an estimate  $w_e$  of w (using the measurement y)?

Reduced-order model of input-output dynamics

$$\begin{cases} \dot{w} = Aw + Bu \\ y = Cw + g \end{cases}$$

Inputs: (u, g)

Output: *y* 



$$y(t) = \int_0^t Ce^{A(t-\tau)}Bu(\tau)d\tau + g(t)$$

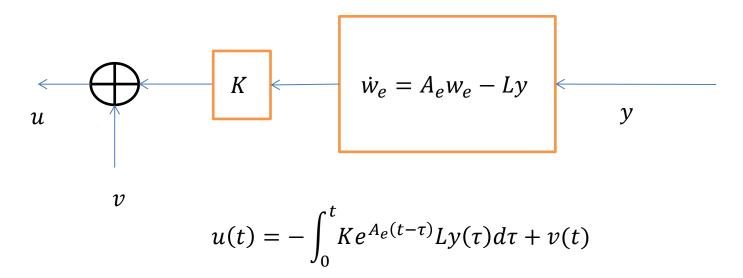
# Observer feedback Compensator

The **compensator** is a system that takes the measurement y and provides a control law u. The controller may be represented as a linear input-output system:

$$\begin{cases} \dot{w}_e = A_e w_e - Ly \\ u = K w_e + v \end{cases}$$

Inputs: the measurement y and the actuator noise v

Output: the control law *u* 



## Observer feedback Dynamic observer

How to choose  $A_e$  and L?

Equation governing system with control input:

$$\begin{cases} \dot{w} = Aw + Bu \\ u = Kw_e + v \text{ (Noisy actuator)} \\ y = Cw + g \text{ (Noisy sensor)} \end{cases}$$

Hence:

$$\dot{w} = Aw + BKw_e + Bv$$

Governing equation of **dynamic observer**: We replace unknown term Bv by a forcing term  $-L(y-y_e)$  proportional to the measurement error:

$$\dot{w}_e = Aw_e + BKw_e - L(y - y_e)$$
$$y_e = Cw_e$$

Hence:

$$\dot{w}_e = (A + BK + LC)w_e - Ly$$

## Observer feedback Dynamic observer

The error  $e = w - w_e$  in the state reconstruction is governed by:

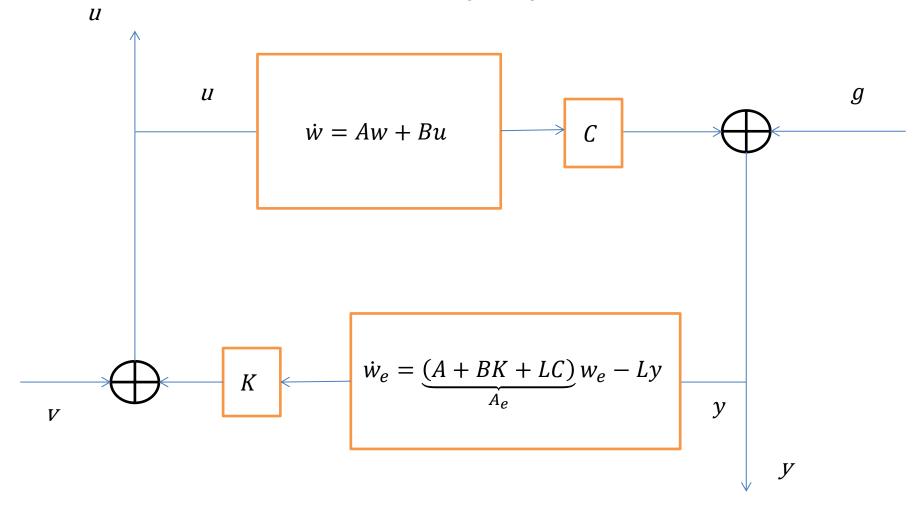
$$\dot{e} = \dot{w} - \dot{w}_e = Aw + BKw_e + Bv - Aw_e - BKw_e + L(Cw + g - Cw_e)$$
$$= (A + LC)e + (B \quad L)\binom{v}{g}$$

We choose L so that A + LC is stable.

$$e(t) = \int_0^t e^{(A+LC)(t-\tau)} [Bv(\tau) + Lg(\tau)] d\tau$$

Hence, error e is weak in presence of noises v and g.

## Observer feedback Closed-loop system



# Observer feedback Closed-loop system

The full coupled system is governed by:

$$\begin{cases} \dot{w} = Aw + Bu \text{ (System dynamics)} \\ \dot{w}_e = (A + BK + LC)w_e - Ly \text{ (Estimator dynamics)} \\ u = Kw_e + v \text{ (Noisy actuator)} \\ y = Cw + g \text{ (Noisy measurement)} \end{cases}$$

Inputs: actuator noise v, measurement noise g

Outputs: measurement y, actuator signal u

In matrix form:

$$\begin{pmatrix} \dot{w} \\ w_e \end{pmatrix} = \overbrace{\begin{pmatrix} A & BK \\ -LC & A + BK + LC \end{pmatrix}}^{A_{cl}} {\begin{pmatrix} w \\ w_e \end{pmatrix}} + {\begin{pmatrix} B & 0 \\ 0 & -L \end{pmatrix}} {\begin{pmatrix} v \\ g \end{pmatrix}}$$

$$\begin{pmatrix} u \\ y \end{pmatrix} = {\begin{pmatrix} 0 & K \\ C & 0 \end{pmatrix}} {\begin{pmatrix} w \\ w_e \end{pmatrix}} + {\begin{pmatrix} v \\ g \end{pmatrix}}$$

# Observer feedback Stability

The dynamics of the coupled system can be analysed by introducing  $e = w - w_e$ :

$$\begin{cases} \dot{w} = Aw + BKw_e + Bv = Aw + BK(w - e) + Bv = (A + BK)w - BKe + Bv \\ \dot{e} = (A + LC)e + Bv + Lg \end{cases}$$

Hence:

$$\begin{pmatrix} \dot{w} \\ e \end{pmatrix} = \begin{pmatrix} A + BK & -BK \\ 0 & A + LC \end{pmatrix} \begin{pmatrix} w \\ e \end{pmatrix} + \begin{pmatrix} B & 0 \\ B & L \end{pmatrix} \begin{pmatrix} v \\ g \end{pmatrix}$$

$$\begin{pmatrix} u \\ y \end{pmatrix} = \begin{pmatrix} K & -K \\ C & 0 \end{pmatrix} \begin{pmatrix} w \\ e \end{pmatrix} + \begin{pmatrix} v \\ g \end{pmatrix}$$

Eigenvalues of coupled system are those of A + BK and A + LC, which by design of K and L, exhibit negative real parts.

# Observer feedback Stability

The compensator is given by:

$$\dot{w}_e = (A + BK + LC)w_e - Ly$$
$$u = Kw_e + v$$

Note that A + BK + LC is not necessarily stable. Only A + BK and A + LC are stable.

## Observer feedback Performance

The performance of the compensator is best when the 2-norm of the closed-loop impulse response is weak. For example, from v to y, this impulse response is:

$$Z^{cl}(t) = (C \quad 0) \exp\left[\begin{pmatrix} A & BK \\ -LC & A + BK + LC \end{pmatrix} t\right] \begin{pmatrix} B \\ 0 \end{pmatrix}$$

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### Laplace transform

#### **Laplace transform:**

$$\hat{f}(s) = \int_0^\infty e^{-st} f(t) dt$$

with  $s = \sigma + i\omega$ .  $\hat{f}(s)$  can be evaluated only if  $\Re(s)$  is sufficiently large.

Laplace transform is analogous to Fourier transform, but also holds for unbounded functions.

#### <u>Inverse Laplace -transform:</u>

$$f(t) = \frac{1}{2\pi i} \lim_{\Omega \to \infty} \int_{\gamma - i\Omega}^{\gamma + i\Omega} e^{st} \hat{f}(s) ds$$

where  $\gamma$  is chosen to the right of all poles of  $\hat{f}(s)$  (causality condition).

### Laplace transform

#### Some useful properties:

$$\begin{aligned} 1/a\widehat{f} + bg &= a\widehat{f} + b\widehat{g} \\ 2/\widehat{f}'(s) &= \int_0^{+\infty} e^{-st} f'(t) dt = \left[ e^{-st} f(t) \right]_0^{\infty} - \int_0^{\infty} -s e^{-st} f(t) dt = \widehat{f}(s) - f(0) \\ 3/H(\widehat{t})e^{at}(s) &= \int_0^{\infty} e^{-st} e^{at} dt = \frac{1}{a-s} \left[ e^{(a-s)t} \right]_0^{\infty} = \frac{1}{s-a} \text{ for } s_r > a_r \\ H(t) \text{ is the Heaviside step function} \\ 4/\widehat{f*g}(s) &= \widehat{f}(s)\widehat{g}(s), \ (f*g)(t) = \int_0^t f(\tau)g(t-\tau)d\tau \\ 5/\operatorname{If} g(t) &= 0 \text{ for } 0 \leq t < \tau \text{ and } g(t) = f(t-\tau) \text{ for } t \geq \tau \text{:} \\ \widehat{g}(s) &= \int_0^{\infty} e^{-st} g(t) dt = \int_{\tau}^{\infty} e^{-st} g(t) dt = \int_{\tau}^{\infty} e^{-st} f(t-\tau) dt \\ &= e^{-s\tau} \int_0^{\infty} e^{-st'} f(t') dt' = e^{-s\tau} \widehat{f}(s) \Rightarrow \arg \widehat{g}(i\omega) = \arg \widehat{f} - \tau \omega \end{aligned}$$

# Frequency space Physical system

Performing a **Laplace-transform** of the governing equations:

$$\begin{cases} \dot{w} = Aw + Bu \\ y = Cw + g \text{ (Noisy sensor)} \end{cases}$$

Yields:

$$\begin{cases}
s\widehat{w} - w(0) = A\widehat{w} + B\widehat{u} \\
\widehat{y} = C\widehat{w} + \widehat{g}
\end{cases}$$

Hence:

$$(sI - A)\widehat{w} = w(0) + B\widehat{u}$$
  
$$\Rightarrow \widehat{y} = C(sI - A)^{-1}w(0) + C(sI - A)^{-1}B\widehat{u} + \widehat{g}$$

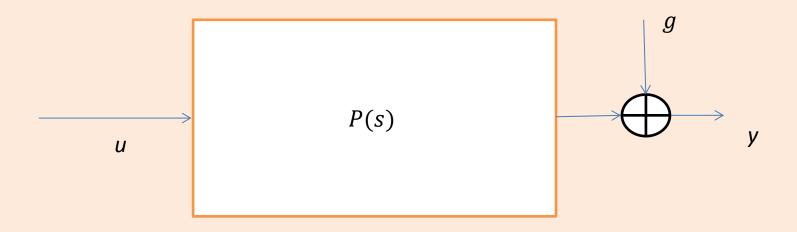
With w(0) = 0:

$$\hat{y} = P(s)\hat{u} + \hat{g}$$

where:

$$P(s) = C(sI - A)^{-1}B$$

### Frequency space Physical system



$$P(s) = C(sI - A)^{-1}B$$

### Frequency space Physical system

The **bode plot** of P(s) presents  $|P(i\omega)|$  and  $\arg P(i\omega)$  as a function of frequency  $\omega$ .

Introducing the adjugate adj(), the transfer function can be rewritten as:

$$P(s) = \frac{C \operatorname{adj}(sI - A)B}{\det(sI - A)} = \frac{\operatorname{num}(s)}{\det(s)}$$

The **poles** of P(s) are defined as the zeros of den(s) and correspond to the eigenvalues of A:  $A\widehat{w} = s\widehat{w} \Rightarrow \det(sI - A) = 0$ .

The **zeros** of P(s) are the zeros of num(s).

### Adjugate

#### Theorem:

$$A^{-1} = \frac{1}{\det A} \operatorname{adj}(A)$$

where adj(A) is the transpose of the matrix of co-factors. For a matrix of order n, the cofactor  $A_{i,j}$  is defined as the determinant of the square matrix of order (n-1) obtained from A by removing the row number i and the column number j multiplied by  $(-1)^{i+j}$ .

Example:

$$\begin{bmatrix} a & b \\ c & d \end{bmatrix}^{-1} = \frac{1}{ad - bc} \begin{bmatrix} d & -c \\ -b & a \end{bmatrix}^* = \frac{1}{ad - bc} \begin{bmatrix} d & -b \\ -c & a \end{bmatrix}$$

# Frequency space Compensator

Taking the Laplace tranform of the equations governing the compenstator:

$$\dot{w}_e = (A + BK + LC)w_e - Ly$$
$$u = Kw_e + v$$

we obtain:

$$s\widehat{w}_e - w_e(0) = (A + BK + LC)\widehat{w}_e - L\widehat{y}$$
  
$$\widehat{u} = K\widehat{w}_e + \widehat{v}$$

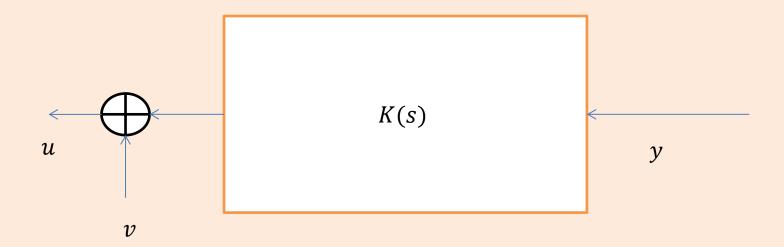
Hence, eliminating  $\widehat{w}_e$  and setting  $w_e(0) = 0$ :

$$\hat{u} = K(s)\hat{y} + \hat{v}$$

where:

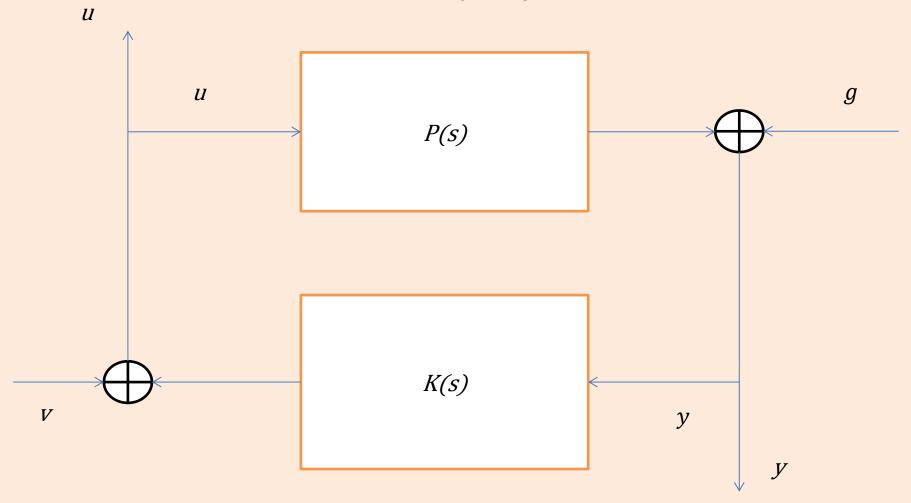
$$K(s) = -K(sI - (A + BK + LC))^{-1}L$$

# Frequency space Compensator



$$K(s) = -K(sI - (A + BK + LC))^{-1}L$$

### Frequency space Closed-loop system



## Frequency space Stability

Closed-loop system is governed by (considering zero initial conditions):

$$\hat{y} = P(s)\hat{u} + \hat{g}$$
$$\hat{u} = K(s)\hat{y} + \hat{v}$$

If we eliminate  $\hat{u}$ , we have:

$$\hat{y} = P(s)K(s)\hat{y} + P(s)\hat{v} + \hat{g}$$

The <u>closed-loop transfer-functions</u> from (g, v) to (y, u) may be obtained from:

$$\hat{y} = \underbrace{\frac{T_{yy}^{cl}(s)}{P(s)}}_{P(s)} \hat{v} + \underbrace{\frac{T_{yg}^{cl}(s)}{1 - P(s)K(s)}}_{K(s)} \hat{g}$$

$$\hat{u} = \underbrace{\frac{1}{1 - P(s)K(s)}}_{T_{uy}^{cl}(s)} \hat{v} + \underbrace{\frac{K(s)}{1 - P(s)K(s)}}_{T_{ug}^{cl}(s)} \hat{g}$$

The **stability of the closed-loop system** is assessed by scrutinizing the poles of the closed-loop transfer-functions. The compensator K(s) is designed to stabilize all of them!

# Frequency space Stability

#### Theorem:

The poles of all closed-loop transfer-functions correspond to the zeros of 1 - P(s)K(s).

#### **Proof:**

Poles of  $T_{yv}^{cl}(s) = \frac{P(s)}{1 - P(s)K(s)}$ : the poles of P(s) in the numerator are cancelled by the poles of P(s) in the denominator. Note also, that the poles of K(s) in the denominator become zeros of the Transfer-function.

Poles of  $T_{yg}^{cl}(s) = T_{uv}^{cl}(s) = \frac{1}{1 - P(s)K(s)}$ . The poles of P(s) and K(s) become zeros of the transfer-function.

Poles of  $T_{ug}^{cl}(s) = \frac{K(s)}{1 - P(s)K(s)}$ : the poles of K(s) in the numerator are cancelled by the poles of K(s) in the denominator. Note also, that the poles of P(s) in the denominator become zeros of the Transfer-function.

### Frequency space Performance

If the closed-loop system is stable, then the **performance** achieved by the compensator is given by the closed-loop transfer functions from v and g to y:

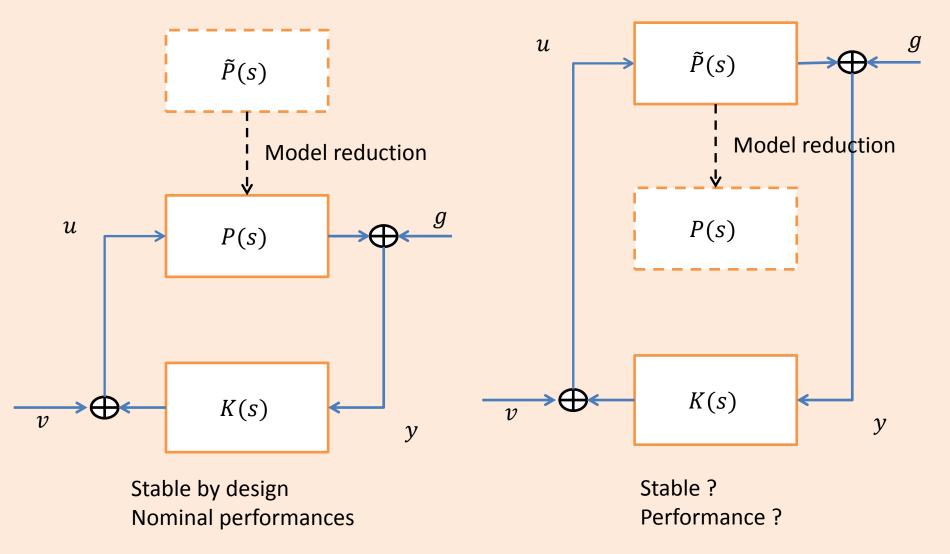
$$T_{yv}^{cl} = \frac{P(s)}{1 - P(s)K(s)}$$
$$T_{yg}^{cl} = \frac{1}{1 - P(s)K(s)}$$

The **control-cost** and the **operating conditions** of the compensator are given by the closed-loop transfer functions from v and g to  $\hat{\mathbf{u}} = K(s)\hat{y} + \hat{v}$ :

$$T_{uv}^{cl}(s) = \frac{1}{1 - P(s)K(s)}$$

$$T_{ug}^{cl}(s) = \frac{K(s)}{1 - P(s)K(s)}$$

### Robustness



### Robustness

The compensator K(s) has been designed based on an estimate of the transfer function: P(s). Yet, the real system may exhibit a slightly different transfer-function:  $\tilde{P}(s)$ .

Nominal performance (performance based on P(s)) is expected when  $\tilde{P}(s) = P(s)$ . When  $\tilde{P}(s) \neq P(s)$ , the actual closed-loop transfer-functions read:

$$\tilde{T}_{uv}^{cl}(s) = \frac{\tilde{P}(s)}{1 - \tilde{P}(s)K(s)}, \qquad \tilde{T}_{ug}^{cl}(s) = \frac{1}{1 - \tilde{P}(s)K(s)}$$

$$\tilde{T}_{uv}^{cl}(s) = \frac{1}{1 - \tilde{P}(s)K(s)}, \qquad \tilde{T}_{ug}^{cl}(s) = \frac{K(s)}{1 - \tilde{P}(s)K(s)}$$

Three things may happen:

- 1/ The actual closed-loop system is stable and exhibits the nominal performances (expected situation)
- 2/ The actual closed-loop system is stable but displays weak performance (bad situation)
- 3/ The actual closed-loop system is unstable: there exists one zero of  $1 \tilde{P}(s)K(s)$  which displays a positive real part (catastrophic situation)

#### Robust controllers

A compensator K(s) displays good <u>stability robustness</u> properties if it stabilizes the closed-loop system for systems  $\tilde{P}(s)$  departing significantly from P(s).

A compensator K(s) displays good <u>performance robustness</u> properties if it stabilizes the closed-loop system and exhibits nominal performance for systems  $\tilde{P}(s)$  departing significantly from P(s).

### Stability robustness analysis

The nominal closed-loop system is stable: the solutions of 1 - P(s)K(s) = 0 all exihibt negative real parts.

We test the stability of the closed-loop system for two families of perturbed transfer functions:

$$\tilde{P}_g(s) = gP(s)$$
  
 $\tilde{P}_{\phi}(s) = e^{i\phi}P(s)$ 

where g and  $\phi$  are real numbers.

Physical interpretation:

- $\tilde{P}_g(s)$  represents an error in the estimate of the growth rate of the instabilities between u and y.
- $\tilde{P}_{\phi}(s)$  represents an error in the estimate of the group velocity of the instabilities between u and y. Note that a time delay is something more complex than just a constant phase-shift!

### Stability robustness analysis

For g=1 and  $\phi=0$ , we have:  $\tilde{P}_g=\tilde{P}_\phi(s)=P(s)$  and the closed-loop system is stable.

We now look for critical parameters g and  $\phi$  which achieve marginal stability, i.e.: there exists  $s=i\omega$  such that  $1-\tilde{P}_g(s)K(s)=0$  or  $1-\tilde{P}_\phi(s)K(s)=0$ . The system is therefore at the threshold of instability.

#### **Definitions:**

1/ the gain margin  $g^+$  is defined as the smallest gain  $\,{\rm g}>1$  , which achieves marginal stability.

2/ the downside gain margin  $g^-$  is the smallest gain  $0 < {\rm g} < 1$ , which achieves marginal stability.

 $3/\phi^+$  is the smallest positive phase shift, which achieves marginal stability.

Note: If  $1 - e^{i\phi^+}P(i\omega)K(i\omega) = 0$ , then  $1 - e^{-i\phi^+}P(-i\omega)K(-i\omega) = 0$  since P(s) and K(s) are polynomials of s with real constants (the matrices (A,B,C) which define P(s) and K(s) are real). Hence,  $\phi^- = -\phi^+$ .

### Stability robustness analysis

A compensator K(s) displays good stability robustness if  $1/g^+$  is large, say  $a^+ \ge 2$   $2/g^-$  is small, say  $a^- \le 0.5$   $3/\phi^+$  is large, say  $\phi^+ \ge 30^\circ$